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# Stability Analysis of Leaning Historic Masonry Structures

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## ABSTRACT

This paper introduces an automatic, powerful and easy to use procedure for undertaking stability analyses of leaning historic masonry structures, based on an upper bound finite element limit analysis (FELA) approach. The procedure proposed here consists of a comprehensive workflow which involves the automatic point cloud manipulation, the 3D mesh generation of the actual geometry for structural purposes (e.g. FE mesh), and a two-step FELA that reduces drastically optimization variables assuming only active few elements inside a restricted processing zone. To generalize the Heyman's intuition to complex real geometries, the use of a 3D upper bound FELA with a recursive kernel of variables reduction becomes necessary for a precise evaluation of the limit inclination that makes the structure collapse under gravity loads. This outcome permits to estimate the structural health condition of a historic structure by comparing the critical inclination angle with the actual one. To demonstrate the effectiveness of the automated procedure, the southwest leaning tower of the Caerphilly castle (Wales, UK) is investigated and failure mechanisms with collapse inclination angles are evaluated through FELA. The proposed procedure presents a high degree of automation at each operational level and, hence, could be effectively used to assess the stability of historic structures at a national scale and provide useful information to asset owners to classify the structural health condition of leaning historic masonry structures in their care.

**Keywords:** Leaning tower; Limit analysis; Point cloud; Mesh generation; Masonry; Actual geometry

## 1 Introduction

Leaning historic masonry structures are fascinating to observe. Perhaps the most famous examples include the Pisa [1, 2] and the Ghirlandina towers [3]. The reason why historic masonry structures lean is a complex area of study and have stimulated the interest of the scientific community for over a century [4, 5, 6, 7]. Due to their narrow foundations, tall and slender historic masonry structures such as towers, whose height is much greater than their width, are generally more prone to lean. Two major reasons why masonry towers tilt are: (a) lack of foundation strength; and (b) lack of foundation stiffness aggravated by progressive soil creep phenomena [4]. Several advanced soil-structure interaction models have been developed to study these phenomena [3]. However, such models require the setting of a numerous amount of mechanical parameters; most of them correlated with in-depth in-situ soil tests. Furthermore, they do not allow for a rapid check on the structural condition of the structure.

Heyman [8] was probably the first to study analytically the safety of leaning towers by assuming masonry as a rigid material unable to withstand tensile stresses. Such simplification allowed deriving a quite simple differential equation describing the crack curve delimiting the failure mechanism and providing very useful hints on the limit inclination angle associated with the collapse of the structure. However, the hypothesis of rectangular full or thin-walled sections and the absence of any irregularity along the height represent a remarkable limitation of the approach, since in practice it is not realistic. Vertical walls of towers vary

considerably in thickness and they often present irregular openings [1]. Furthermore, historic towers, or other height-prevalent historic structures (e.g. walls in churches, curtain-defensive walls etc.), frequently stand in a ruined condition, and have been subjected to unforeseen load events (e.g. bombing, successive demolitions, sabotages, raids etc.) over the centuries [9]. Often, such structures suffered alternations and today, only a few portions of the original structure remain standing. Consequently, the geometry of these structures is generally extremely complex and irregular (see for instance [6]). Their complex geometries suggest to consider advanced methods of analysis where the actual 3D geometry of the structure is accounted for in the calculations [10].

One of the first challenging tasks that appears when dealing with the numerical modelling of this kind of structures lasts in acquiring their 3D geometric features. The use of automated surveying techniques, such as terrestrial laser scanning [11] and close-range photogrammetry [12] which produce dense point clouds, appears particularly suitable for obtaining the geometry of historic structures [13, 14, 15]. From such studies, it is evident that the use of terrestrial laser scanning and close-range photogrammetry surveying techniques could result in accurate representation of the geometry of a structure in a relatively reduced cost [16, 17, 18, 19]. Documentation for supporting restoration works as well as monitoring of historic structures are common motivations for the use of these surveying techniques on structures of architectural heritage [20, 21, 22, 23].

Nevertheless, the exploitability of laser scanning and photogrammetric surveys outputs, i.e. point clouds, for structural analysis of historic building is still challenging. A significant contribution to this field is proposed in [24], where an attempt to accurately capture the geometry of a structure by automatic reconstruction is presented. Moreover, in [25], a point-based voxelization method for automatic transformation of point cloud data into solid models for computational modelling was proposed. The approach involves the construction of a triangular irregular network (TIN) mesh by means of a voxel grid bounding the cloud region. However, such approach is limited to building façades and it does not capture the entire geometrical domain of the structure.

Dealing with actual buildings, two main difficulties arise: (i) the automatic filling of very long vacancies (roof structures for instance), and (ii) reduction of the whole model in a more simplified and compressed form. To overcome these issues, a semi-automatic procedure (called CLOUD2FEM) to transform three-dimensional point clouds of complex objects into a three-dimensional finite element model has been presented in [26]. The procedure conceives of the point cloud as a stacking of point sections and aims at solving the problems connected to the generation of finite element models of these complex structures by constructing a fine discretized geometry with a reduced amount of time and outputs ready to be used for structural analysis problems [27]. A recent application of this procedure to a full-scale medieval fortress and its subsequent seismic assessment by means of nonlinear static analysis is reported in [28].

Once the mesh is available, it can be used within several computational tools for the structural analysis of historic masonry buildings. Particularly, ~~these-such~~ approaches are usually based on: a) the finite element method (FEM) [29, 30, 31, 32, 33, 28, 34, 35]; b) the discrete element method (DEM) [36, 37]; and c) limit analysis [38, 39]. Interesting comparisons of computational techniques are provided in [40, 41, 42].

Another interesting issue is to determine the maximum inclination angle that leads to the collapse of a tower due to the loss of stability under the application of gravity loads. Concerning a no-tension material with rectangular cross section, Heyman [8] was able to analytically determine the collapse inclination angle and the corresponding crack pattern shape. Unfortunately, an analytical approach for real irregular geometries is hardly applicable, and an automatic procedure is needed. To deal with such key issue is one of the two aims of the present paper.

Firstly, a simplified and rapid procedure for the automatic transformation of point clouds (surveyed on historic structures) to 3D FE meshes, passing from the concept of watertight mesh, is proposed. The accuracy of the geometry of the mesh generated appears suitable for structural purposes. Secondly, following the Heyman's work [8], the use of a two-step 3D upper bound finite element limit analysis (FELA) on the generated mesh is conducted ~~and to evaluate~~ the critical condition (i.e. maximum inclination capacity) of a leaning historic masonry structure is evaluated.

The procedure is indeed an upper bound limit analysis with FE discretization obtained by means of tetrahedron rigid elements and rigid-perfectly plastic interfaces exhibiting frictional behaviour and very low cohesion (i.e. mimicking a quasi no-tension material with friction).

Considering that FE meshes obtained from detailed laser scanner surveys would be constituted by hundreds of thousands of elements and interfaces, the limit analysis problems derived would be characterized by millions of variables, i.e. in practice impossible to solve even with super-computers. An alternative to parallelization is proposed for the first time here, which is essentially a master-slave approach conceived with the aim of reducing drastically the total number of optimization variables. The procedure is based on the hypothesis that the tower collapses for the plasticization of few elements located in a limited processing zone, which is a-priori established in the first step. Elements with centroids inside the processing zone are assumed potentially active. The rest of the mesh is excluded from computations and it is treated as a single rigid body characterized by six degrees of freedom (i.e. three centroid velocities and three rotation rates). The solution of the linear programming problem found in the first step provides a more accurate estimation of the potential interfaces undergoing plasticization. In the second step, the processing zone is further reduced to those elements whose interfaces exhibit meaningful inelastic deformation rates plus few contiguous ones, to further drop-down the optimization variables. Conversely, the failure surface linearization on the active interfaces is refined to obtain more accurate estimates of the collapse multiplier. The master-slave kernel is then coupled with a sequential linear programming algorithm to deal with the linearization of the normalization condition equation, which results nonlinear due to the assumption of the inclination angle at failure as collapse multiplier.

The outcome obtained with the limit analysis permits to practically estimate the structural health condition of a leaning historic structure, e.g. by comparing the maximum critical inclination angle ~~with~~<sup>against</sup> the actual one. Since the computational approach proposed herein presents a high degree of automation at each operational level (i.e. survey, point cloud manipulation, mesh generation, numerical analysis), its usage could be addressed to the stability analysis of historic structures at a national scale. In particular, this approach could be ~~very~~ beneficial for asset managers, which want to classify the structural condition of leaning historic assets in their care and devise action plans for their survival.

In this research, the southwest leaning ruined masonry tower of the Caerphilly castle (Wales, UK) is employed as a case study to demonstrate the effectiveness of the proposed approach. The paper is organized as follows. Section 2 briefly presents the case study. Section 3 describes the automatic procedure for the stability analysis of leaning historic structures. Section 4 reports the analyses results and their discussion. Finally, Section 5 highlights the main conclusions of this research work.

## 2 The leaning tower of Caerphilly castle, UK

In this section, a brief description of the case study related to the southwest leaning tower of Caerphilly castle, used to test the effectiveness of the proposed approach is reported. The case study is merely used to demonstrate the effectiveness of the proposed automatic procedure for assessing the structural stability of leaning historic masonry structures.

Caerphilly castle is a medieval fortification in Caerphilly, South Wales, UK. The castle was constructed by Gilbert de Clare in the 13th century [43] and it is the second largest in the UK. [Fig. 1](#) shows the southwest leaning ruined tower of the Caerphilly castle. The tower has been in a ruined and leaning condition for several centuries [8, 44].



Fig. 1 – Southwest tower of Caerphilly castle: a) historic drawing (A.D. 1773) and b,c) photos of the tower in its actual condition.

The tower is approximately 17 m tall. It used to have a circular ground plant of approximately 9 m in diameter. Today, the inclination of the tower is approximately 10 degrees. For the sake of comparison, the campanile of Pisa is 55.86 m tall and leans at an angle of 5.5 degrees. The southwest tower is made of stone masonry with a fully irregular texture. No information is available about masonry material properties, soil stratigraphy and foundations. However, medieval fortified structures were generally characterized by particularly shallow foundations [9]. According to Renn [43], the deterioration subsequent leaning of the tower was probably the result of subsidence caused by dewatering in the 18<sup>th</sup> century, as there is no evidence of deliberate destruction having been ordered.

In 2014, a detailed survey of the southwest tower commissioned to document its health condition [44]. A total of 27 scans were surveyed using a FARO focus 3D X130 terrestrial laser scanner. The inclinometer, altimeter, compass, clear contour and clear sky were also activated and far distance deactivated. A series of scans have been taken from points on the ground around the base of the tower. During the scanning, challenges were mainly due to scaffolded areas around the tower and the presence of tourists. The complexity of the geometry also proved challenging in the placement of targets. Twelve spherical targets used to locate the scans. The resulting points' cloud is shown in [Fig. 3](#).

### 3 The procedure

In this section, the automatic procedure for the stability analysis of leaning historic structures is described. It comprises the automatic FE mesh generation starting from point cloud, the 3D limit analysis model description and its utilization for the stability analysis of masonry structures. The case study presented in the previous section is used to explicitly show the main steps of the procedure.

#### 3.1 FE mesh generation

In the following, a simple [approach for](#) processing ~~of~~ point clouds is suggested for the automatic transformation of point clouds (surveyed on historic structures) to 3D FE meshes, following the concept of watertight mesh. Watertight means that the mesh on all of the surfaces is complete, so that the volume is fully enclosed. This approach appears appealing for historic ruined structures, which are generally characterized by the absence of inner spaces, furniture, doors and window frames etc. The proposed mesh generation approach treats historical structures similar to 3D objects such as stone sculptures, building ornamental components [45], etc and follows an akin concept of digital reproduction of 3D objects from 3D scanning to 3D printing [45]. However, in this instance, instead of using the mesh to perform 3D printing, its usage is aimed ~~to for~~ structural analysis purposes (i.e. FE mesh made of 3D continuum elements).



The proposed procedure requires a point cloud of the historic structure under study. The point cloud could be the output of either terrestrial laser scanning or close-range photogrammetry. These survey techniques generate dense clouds of 3D points, generally characterized by several millions of points for full-scale buildings.

Often, a few preliminary and standardized operations on the point cloud dataset may be required to facilitate the subsequent operations. Firstly, a specific algorithm [46] is used to populate a new dataset of points with a sampling generated according to a Poisson-disk distribution. The resulting reduced point cloud is characterized by a regular spatial sampling, which can be chosen depending on the level of detail desired for the case at hand. Dealing with historic structures, a point cloud density between 10,000 and 40,000 pts/m<sup>2</sup> results a good compromise between the manageability of the dataset and the accuracy in the representation of the geometry for structural purposes [26]. Indeed, millimetric details are generally negligible to analyse the structural behaviour of historic constructions. Successively, the point cloud has to be cleaned mainly by removing all neighbour points, which are not relevant to the structure. Although this operation is manual, it is easy to implement and substantially fast.

Afterward, a TIN mesh is generated based on the point cloud (in the condition after sampling and cleaning). The TIN mesh is created by linking triplets of nodes to form non-overlapped triangles [47]. Ideally, if every portion of the object's surface is meticulously surveyed, this mesh could be already watertight. However, this never occurs for large-scale and complex structures since their surfaces are continuous with the terrain. Furthermore, few parts of the object's surface are commonly missing, due to access difficulties during the survey.

Since the aim is to obtain a watertight mesh, further processing of the data is necessary. Thus, the Poisson surface reconstruction [48] can be adopted. Poisson surface reconstruction is a well-known computer graphics technique for creating watertight surfaces from oriented point samples acquired with 3D range scanners. This algorithm expresses surface reconstruction as the solution to a Poisson equation. Therefore, Poisson reconstruction considers all the data at once, creating very smooth surfaces that robustly approximate noisy data [48], suffering, however, from a tendency to over-smooth the data. All these operations are basically common and standardized, and they can be conducted by means of any 3D mesh processing software.

Once the watertight mesh has been obtained, it can be directly exploited for setting up the 3D FE mesh. The main aim is to generate a final mesh that will be robust when used for finite element calculations. Although an adaptive (skin) mesh could be used in areas where more detail is desired, it is even more important to have a final (solid) mesh with appropriate topology. To this aim, three alternative operations are suggested herein.

In the first case, the triangles which constitute the watertight mesh are directly transformed into FE triangles. Then, the triangles are transformed into solid 4-nodes tetrahedral FEs filling the whole volume by means of automatic advancing front methods which are already implemented in most commercial software packages (see for example the one implemented in Abaqus [49]). This operation is fully automatic, and therefore, extremely fast. In addition, it does not introduce any further geometric approximation. Nevertheless, being the watertight mesh not conceived for FE analysis, this approach could lead to excessively distorted finite elements and, therefore, it could fail. In most cases, few local manual refinements could overcome this drawback, but sometimes the simple regularization of the metrics on the surface (watertight mesh) cannot lead to a good (undistorted) solid mesh. This aspect mainly depends upon-on the mesh generation software that can or cannot employ robust mesh generation procedures.

In the second case, the volume defined by the watertight mesh is filled by voxel, using well-known and established voxelization algorithms (see for instance [50]). The outcome strictly relies on the adopted voxel dimension, which can be chosen depending on the case at hand (possibly comprised in between 0.05 and 0.25 m for historic structures [26]). In this case, the voxel model is nothing more than a 3D matrix in which the value 1 refers to the structure and the value 0 represents voids, as well as the information of the voxel dimensions. Therefore, their transformation to 8-nodes hexahedral FEs is a simple, common and standardized operation [26]. The voxel dimension should represent a good compromise between geometric accuracy and computational effort of FE analyses. Given a watertight mesh, its voxelization is always possible and the

generation of the FE mesh is guaranteed. Therefore, although the voxelized model is generally rougher than the one obtained through direct transformation of the watertight mesh, the voxelization of the domain represents a robust and always guaranteed approach for the FE mesh generation.

In the latter case, the watertight mesh is processed by means of a retopology algorithm (see for example [51]) to generate a coarser representation of the surface [52]. This kind of algorithms are able to re-mesh a surface into a triangular mesh, with a certain resolution chosen by the user, using unified local smoothing operators that optimize both the edge orientations and vertex positions in the output mesh [51]. Then, the triangles are transformed into solid 4-nodes tetrahedral FEs. Therefore, the problem dimensions, as well as the number of solid FEs, are broken down thanks to the reduction of the triangles of the watertight mesh. This operation produces a further approximation of the surface depending on the resolution of the triangles chosen.

In ~~the a~~ case ~~when-where~~ the surveyed structure presents several inner spaces (e.g. rooms) and the presence of furniture, doors and windows, i.e. the TIN mesh cannot become watertight, other semi-automatic mesh generation approaches can be employed. The interest reader is referred to the CLOUD2FEM procedure, recently developed by part of the authors of this manuscript [26, 27, 28]. Indeed, although this procedure is more laborious than the proposed one, being not completely automatic, it always guarantees the mesh generation of the 3D geometry. It has to be pointed out that the transformation of hexahedral FEs into tetrahedral FEs is always possible and it is a very common operation, see for instance [53]. In ~~Fig. 2~~[Fig. 2](#), a general flowchart for the FE mesh generation from point clouds is shown.





### 3.1.1 FE mesh generation of the case study

[Fig. 3Fig. 3](#) presents the main outcomes of the mesh generation operations conducted on the case study. [Fig. 3Fig. 3a](#) shows the initial point cloud and [Fig. 3Fig. 3b-c](#) shows the TIN mesh. As it can be observed, several portions of the structure surface are lacking due to the aforementioned drawbacks in the surveying operations. However, the main geometric features of the structure were successfully collected. The Poisson surface reconstruction has been conducted on the case study's data set, characterized by a point cloud density equal to 20,000 pts/m<sup>2</sup>. Although no significant variation of the mesh quality has been observed with point cloud densities between 10,000 and 40,000 pts/m<sup>2</sup>, substantially finer values (e.g. 400,000 pts/m<sup>2</sup>) led to excessively large meshes which were unable to be processed, whereas considerably coarser values (e.g. 1000 pts/m<sup>2</sup>) led to an excessive approximation of the geometry. [Fig. 3Fig. 3d-e](#) shows the resulting watertight mesh consisting of triangles, where it is superimposed on the TIN mesh of the previous step. As can be noted, the watertight mesh reasonably approximates the original TIN mesh. In this case, the open source mesh processing tool MeshLab [54] has been utilized. The level of approximation introduced in the geometry in this circumstance appears to be included, in the authors' opinion, within the engineering tolerance.

On the one hand, the FE model of the case study obtained through voxelization is shown in [Fig. 3Fig. 3f](#), where the value 0.175 m has been adopted as voxel dimension. On the other hand, [Fig. 3Fig. 3g](#) shows the solid FE model obtained through the direct transformation. In this case, no excessive distortion arose. Therefore, this latter model has been herein preferred over the voxelized model since it does not introduce further geometric approximations, as voxelization does. [Fig. 3Fig. 3h](#) shows the mesh used for the analyses. The mesh consists of 28,738 nodes and 145,893 4-nodes tetrahedral FEs. All the nodes of the bottom part of the mesh have been projected to a horizontal plane to allow the easy setup of the boundary conditions in the model ([Fig. 3Fig. 3h](#)). Finally, the solid mesh ([Fig. 3Fig. 3i](#)) obtained through retopology [51] (with an average size of the side of the triangles equal to 60cm) consists of 5,639 nodes and 27,114 tetrahedral FEs. As can be noted, the mesh is characterized by a largely reduced number of solid FEs. Nevertheless, it is still suitable for structural purposes.

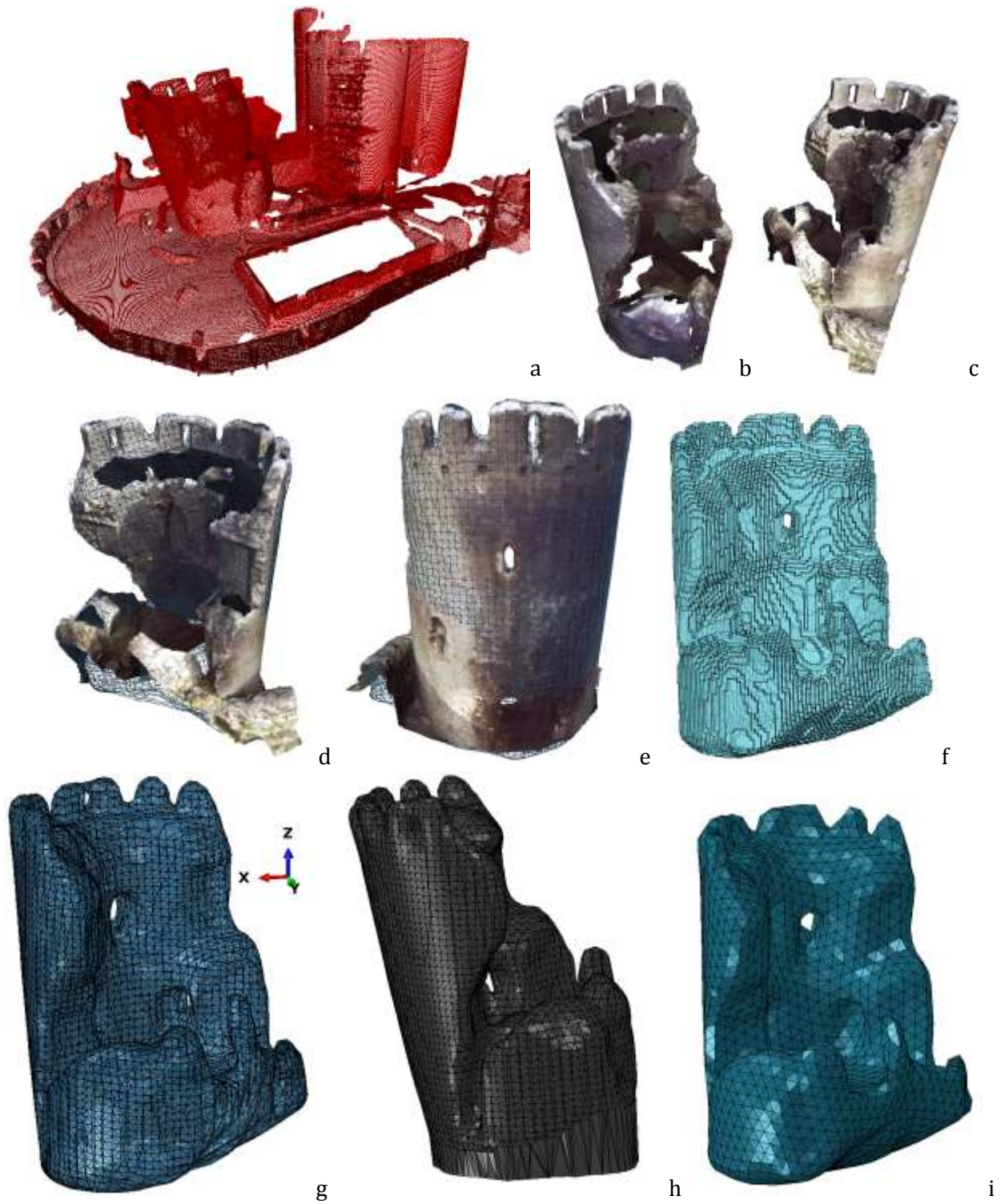


Fig. 3 – Points cloud manipulation and mesh generation: a) rough points cloud, b-c) tower's TIN mesh, d-e) superposition of TIN mesh and watertight mesh, f) example of voxelization of the domain, g) transformation of the watertight mesh into a tetrahedral mesh, h) mesh used for numerical analyses and i) solid mesh obtained through retopology.

### 3.2 Limit analysis for the stability assessment of 3D leaning masonry structures

The stability assessment of 3D masonry structures having complex geometry by means of limit analysis can be performed with 4-nodes rigid infinitely resistant tetrahedrons and triangular interfaces where plastic dissipation can occur. The utilization of interfaces has proved to be effective for cohesive frictional materials [55, 56, 57, 58], even if yield lines are fixed and re-meshing would be required. However, the utilization of rigid elements reduces to a great extent the optimization variables, which is desired in case of complex geometries. The following formulation permits the flow rule to be violated in the discontinuities preserving, nevertheless, the upper bound property of the solution. The discontinuity occurs at the common edge between two adjacent tetrahedrons. To be kinematically admissible, the normal and tangential velocity jumps across the discontinuity must satisfy the flow rule.

#### 3.2.1 Finite element limit analysis 3D model

The kinematic variables of a tetrahedron  $E$  are the three centroid velocities  $(u_x^E, u_y^E, u_z^E)$  and the three rotation rates  $(\Phi_x^E, \Phi_y^E, \Phi_z^E)$  around the centroid (Fig. 4Fig. 4). Jump of velocities  $[\mathbf{U}(P)]$  at a point  $P = (x_P, y_P, z_P)$  on an interface  $I$  between two adjoining elements  $M$  and  $N$  are evaluated as follows:

$$[\mathbf{U}_P^M - \mathbf{U}_P^N] = \mathbf{U}_G^M - \mathbf{U}_G^N + \mathbf{R}^M(P - G^M) - \mathbf{R}^N(P - G^N)$$

$$\mathbf{U}_P^E = \begin{bmatrix} u_x \\ u_y \\ u_z \end{bmatrix} = \begin{bmatrix} u_x^E \\ u_y^E \\ u_z^E \end{bmatrix} + \begin{bmatrix} 0 & -\Phi_y^E & \Phi_z^E \\ \Phi_y^E & 0 & -\Phi_x^E \\ -\Phi_z^E & \Phi_x^E & 0 \end{bmatrix} \begin{bmatrix} x_P - x_G \\ y_P - y_G \\ z_P - z_G \end{bmatrix} = \mathbf{U}_G^E + \mathbf{R}^E(P - G) \quad (1)$$

$[\mathbf{U}(P)]$  must be written in the local coordinate system of the interface  $\mathbf{r}_1^I - \mathbf{r}_2^I - \mathbf{s}^I$  as  $\Delta \mathbf{U}(P) = [\Delta r_1 \quad \Delta r_2 \quad \Delta s]^T = \mathbf{R}^I[\mathbf{U}(P)]$ , where  $\Delta r_1$ ,  $\Delta r_2$  and  $\Delta s$  are the velocity jumps (two tangential and mutually orthogonal and one perpendicular to the interface, see Fig. 4Fig. 4) in the local coordinate system and  $\mathbf{R}^I$  is the rotation matrix of the local frame of reference with respect to the global one.

Let's  $\mathbf{t}^{I^T} = [\tau_1^I \quad \tau_2^I \quad \sigma_s^I]$  denotes the stress vector with normal and tangential components acting along local axes  $\mathbf{r}_1^I(\tau_1^I)$ ,  $\mathbf{r}_2^I(\tau_2^I)$  and  $\mathbf{s}^I(\sigma_s^I)$ , see Fig. 4Fig. 4.

On the interface  $I$  of the area  $\Omega$  connecting nodes 1-2-3 (Fig. 4Fig. 4), assuming for the masonry material a linearized strength domain in the local coordinate system constituted by  $m^I$  planes ( $q^I$  plane of equation  $A_{r_1}^{q^I} \tau_1^I + A_{r_2}^{q^I} \tau_2^I + A_{\sigma_s}^{q^I} \sigma_s^I = C_I^{q^I}$   $1 \leq q^I \leq m^I$ ), see Fig. 5Fig. 5, and introducing fields of plastic multipliers at the interface (one for each linearization plane), it can be easily shown that power dissipated is the following:

$$P^I = \frac{\Omega}{3} \sum_{q^I}^{m^I} (\lambda_{q^I}^{I,1} + \lambda_{q^I}^{I,2} + \lambda_{q^I}^{I,3}) C_I^{q^I} \quad (2)$$

where  $\lambda_{q^I}^{I,j}$  is the  $q^I$  plastic multiplier of node  $j$ .

Associated flow rule constraints are imposed again on interface nodes in the following form:

$$\Delta \mathbf{U}(P_j) = \sum_{q^I}^{m^I} \lambda_{q^I}^{I,j} [A_{r_1}^{q^I} \quad A_{r_2}^{q^I} \quad A_{\sigma_s}^{q^I}]^T \quad (3)$$

The external power dissipated can be written as  $P^{ex} = (\mathbf{P}_0^T + \lambda \mathbf{P}_1^T) \mathbf{v}$ , where  $\mathbf{P}_0$  is the vector of permanent loads,  $\lambda$  is the load multiplier for the structure examined,  $\mathbf{P}_1$  is the vector of variable loads and  $\mathbf{v}$  is the vector of assembled centroid elements velocities. As the amplitude of the failure mechanism is arbitrary, the additional classic normalization condition  $\mathbf{P}_1^T \mathbf{v} = 1$  is also added, reducing thus the external power to linearity.

Formally, the linear programming problem obtained is classic, and relies into the constrained minimization of the total internal power dissipated minus the power dissipated by external loads which do not depend by  $\lambda$ :

$$\begin{cases} \min \{ \mathbf{P}_I^{in,ass} \dot{\lambda}^{I,assT} - \mathbf{P}_0^T \mathbf{v} \} \\ \text{such that } \begin{cases} \mathbf{A}^{eq} \mathbf{U} = \mathbf{b}^{eq} \\ \dot{\lambda}^{I,ass} \geq 0 \end{cases} \end{cases} \quad (4)$$

where  $\mathbf{U}$  is the vector of global unknowns which collects the vector of elements centroids velocities ( $\mathbf{v}$ ) and rotations ( $\Phi$ ) of masonry elements and the vector of assembled interface plastic multiplier rates ( $\dot{\lambda}^{I,ass}$ ).  $\dot{\lambda}^{I,ass}$  is the vector of assembled plastic multipliers of the interfaces. Also,  $\mathbf{A}^{eq}$  and  $\mathbf{b}^{eq}$  are the overall constraints matrix and vector and collect normalization condition, velocity boundary conditions and plasticity normality rules on interfaces.

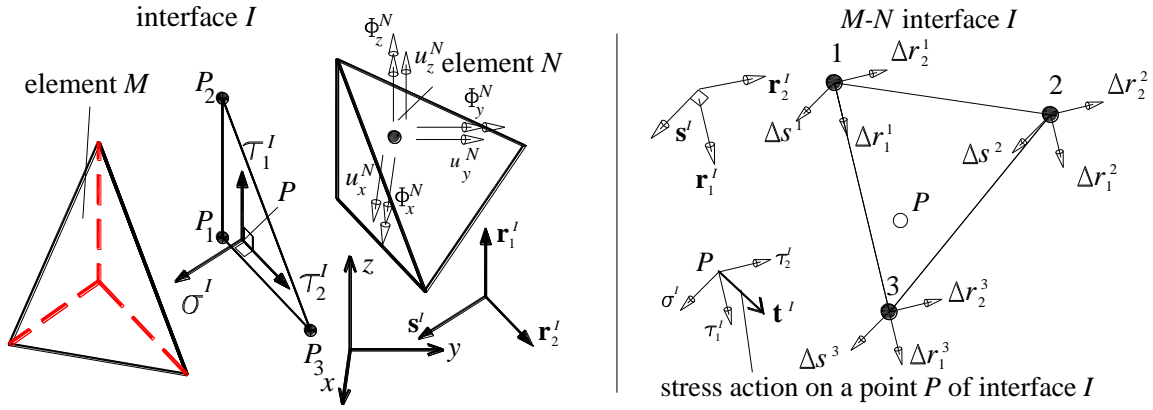


Fig. 4 – Interface  $I$  between two adjoining elements  $M$  and  $N$ , stress acting on the interface and local frame of reference (left); jump of velocities of nodes of the same interface in the local frame of reference (right).

### 3.2.2 Master-Slave elements: **H**ow to manage problems with over one million degrees of freedom (dof)

Aiming at reducing the numerous amount of optimization variables involved in the limit analysis computations, a master-slave approach is herein proposed. Indeed, considering that the FE model studied is constituted by 145,893 elements and approximately 170,000 interfaces and assuming rigid elements (i.e. those with six degrees of freedom to determine) and interfaces obeying a rough linearization of the Mohr-Coulomb failure criterion with tension cut-off with only 5 plastic multipliers (at least in the first step, as it will be explained later on), the limit analysis problem derived would be characterized by slightly less than 2 million variables. This is certainly a linear programming problem which requires the utilization of either super-computers or parallelization, with considerably long computational times or out-of-memory issues to tackle. In order to circumvent such problems and reduce the huge computational effort required, the two-step procedure shown in Fig. 5 is adopted. Substantially, a rough approximation of the Mohr-Coulomb failure surface with tension cut-off is utilized in Step 1 to reduce the computational burden. This approximation is constituted by only 5 planes. Step 1 is used to roughly identify the active failure mechanism and to proceed to Step 2 using a new active zone band, closer to the actual one. In Step 2, a more refined discretization of the failure domain is utilized to determine with higher accuracy the active failure surfaces inside each interface.

In the Step 1, a processing zone is a-priori established, assuming only active interfaces between two distances  $z_1$  and  $z_2$  from the ground. All elements with centroids inside such region are assumed potentially active and

any interface involving at least one of such elements is also considered active. Regions outside the interval  $z_1$ - $z_2$  are assumed to behave kinematically as a rigid block, fully characterized by the knowledge of the position of the centroid and six degrees of freedom (3 rotations around the centroid and 3 velocities of the centroid). All interfaces outside the established processing zone are indeed not active and contiguous elements cannot separate in any manner. Being rigid elements, they are therefore constrained to move as part of the same rigid body. Therefore, the slave region is treated as the assemblage of few single rigid elements with complex geometry, so the finite element nodes on the edges between the meshed region and the rigid bodies are the slave nodes being dependent on the degrees of freedom expressed at the centroids of the rigid bodies.

Although this assumption is arbitrary, the user generally knows in advance the nature of the expected failure mechanism. For the problem under investigation, a failure mechanism characterized by a shear-flexural hinge located near the base is expected. The exclusion of the mesh on the upper part does not preclude in any manner a correct estimation of failure loads and collapse mechanisms. In Step 1, there is also no need to refine the linearization of the failure surface assumed for the interfaces, because the aim is only to identify possible sub-regions of active interfaces. In this manner, the obtained linear programming problem contains a reasonable total number of optimization unknowns. The width of the active region can be properly tuned to obtain problems that are computationally more sustainable for the **available computers**.

Once the exact regions of active interfaces are known (from the solution of the linear programming problem in Step 1), the processing zone is again reduced in Step 2 to further drop-down the optimization variables. Generally, authors experienced that such new processing zone found in Step 1 should be slightly enlarged including 2 or 3 rows of neighbouring inactive elements with the aim of compensating potential inaccuracies of the procedure due to, for example, the rough discretization adopted for the failure surfaces of the interfaces in Step 1. The strong reduction of the active interfaces in Step 2 allows for a refinement of the linearization of the interfaces failure surfaces, with the possible inclusion of a compression cut-off, although this is not essential for such kind of problems.

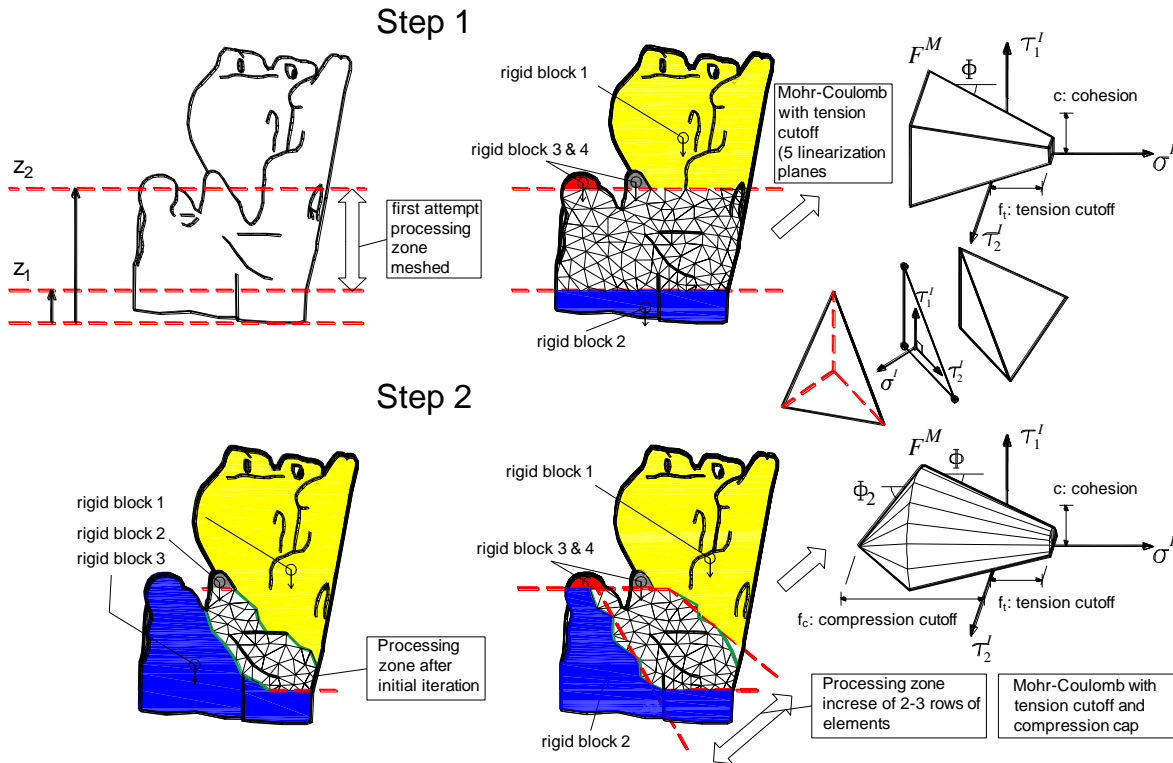


Fig. 5 – Two-step master-slave approach adopted.



In Step 1, computations are performed assuming 129,872 slave elements, 4 master regions of rigid blocks and a total number of active elements equal to 16,021, as depicted in [Fig. 6](#) (active elements are indicated in orange in the picture). Such a choice leads to tackle a linear programming problem constituted by about 14,000 interfaces and around 160,000 optimization variables. As it will be discussed later on in the paper, the evaluation of the ultimate inclination can be assimilated to the study at collapse of the same structure resting on a tilting plane. Such problem becomes nonlinear, due to the presence of a few nonlinearities. A sequential linear programming (SLP) strategy is therefore adopted, linearizing the nonlinear constraints and repeating a sequence of few linear programming iterations, typically 4-5. On a work station equipped with 32 in-parallel processors and CPLEX solver [59], Step 1 within the SLP procedure is solved in less than 3 hours. The rough identification of the processing zone reduces further the number of active elements to about 4,000, with less than 5,000 active interfaces. Assuming a quite refined linearization of the failure surface for the interfaces with 25 planes, the total number of optimization variables involved is roughly 150,000, comparable with those utilized in Step 1 and hence, with similar processing times needed.

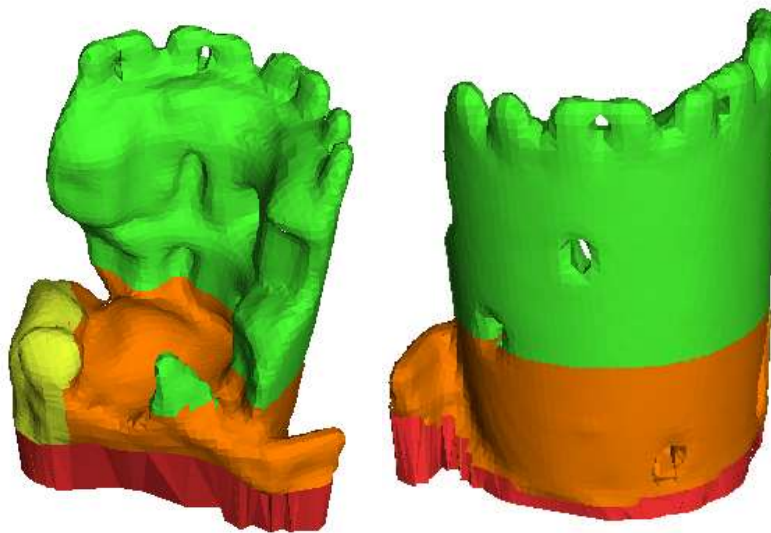


Fig. 6 – Slave elements: 129,872, Master regions: 4, Total #elements: 16,021.

### 3.2.3 Limit analyses on a tilting plane

The limit inclination angle of a masonry structure (i.e. the angle which leads the collapse of the structure under gravity loads only) can be computed within a mathematical programming FE procedure. This is a particular type of limit state problem, where the collapse of a structure resting on a tilting plane is determined by progressively increasing the inclination of the base plane from the horizontal direction (i.e. by increasing the angle  $\vartheta$ , see [Fig. 7](#)) up to collapse. In this framework, the collapse multiplier adopted is  $\vartheta$ , i.e. the limit inclination angle of the tilting base.

External loads (due exclusively to gravity) are the components of the gravity load which act perpendicular and tangential to the tilting base. The progressive rotation of the tilting plane results into a decrease of the vertical load and an increase of the horizontal one, according to the decomposition of the gravity loads  $\mathbf{W}$  into the horizontal component ( $\mathbf{W} \sin \vartheta$ ) and the vertical component ( $\mathbf{W} \cos \vartheta$ ) respectively, [Fig. 7](#). If the inclination angle of the tilting plane  $\vartheta$  is assumed to play the role of the collapse multiplier in the limit analysis problem ([4-4](#)), it is worth noting that there is no way to deal with it using linear programming routines, because the normalization condition becomes nonlinear. Obviously, the external power dissipated (which contains only loads dependent upon the collapse multiplier) is nonlinear. In limit analysis, loads are applied contemporarily at the same time and a snapshot of the situation is provided at failure via the solution of a constrained minimization problem.



In such a particular case,  $P^{ex} = \left( P_0^T + \vartheta \frac{P_1^T(\vartheta)}{\vartheta} \right) \mathbf{v}$ , where again  $P_0$  stands for the vector of loads not dependent on  $\vartheta$ ,  $P_1^T$  is the vector of all loads dependent on  $\vartheta$  and  $\mathbf{v}$  is the vector of assembled velocities at the centroid of the elements. The normalization condition is obviously  $\frac{P_1^T(\vartheta)}{\vartheta} \mathbf{v} = 1$ , which is clearly a nonlinear function.

Such nonlinearity has the consequence that the obtained mathematical programming problem is nonlinear. Almost identical to (4-4), such nonlinear programming problem -where the nonlinear normalization condition is put in evidence-, can be written as follows:

$$\begin{cases} \min \{ \mathbf{P}_I^{in,ass} \dot{\lambda}^{I,assT} - \mathbf{P}_0^T \mathbf{v} \} \\ \text{such that} \begin{cases} \mathbf{A}^{eq} \mathbf{U} = \mathbf{b}^{eq} \\ \frac{P_1^T(\vartheta)}{\vartheta} \mathbf{v} - 1 = 0 \\ \dot{\lambda}^{I,ass} \geq 0 \end{cases} \end{cases} \quad (5)$$

The nonlinear programming problem (5-5) can be solved by means of two distinct possible approaches: a) using a standard non-linear programming (NLP) routines, and b) implementing a sequential linear programming SLP procedure, linearizing the nonlinear normalization condition. This last approach is preferred here, and it is worth mentioning that the linearization of the nonlinear equation can be written as follows:

$$\left( \frac{P_1^T(\vartheta_0)}{\vartheta_0} \mathbf{v}_0 - 1 \right) + \frac{\frac{dP_1^T(\vartheta)}{d\vartheta} \vartheta \mathbf{v} - P_1^T(\vartheta) \mathbf{v}}{\vartheta^2} \bigg|_{\substack{\vartheta=\vartheta_0 \\ \mathbf{v}=\mathbf{v}_0}} \vartheta = 0 \quad (6)$$

Where subscript 0 indicates the solution of the previous iteration. The starting point selected at the beginning of the SLP procedure when linearizing Eq. (6)-(6) coincides with an inclination angle of the tilting plane equal to  $\vartheta = 0$ . Therefore, the limit inclination angle of the structure is evaluated as the sum of the limit inclination angle of the tilting plane  $\vartheta$  and of the actual inclination of the structure. As a matter of fact, the evaluation of the actual inclination of the structure is not an easy task, because it depends on the in-plane direction along which the inclination is evaluated. For the case study at hand, plotting the in-plane position of nodes belonging to two horizontal planes as shown in Fig. 8, and assuming that the vertical distance between the two planes is known, it is possible to estimate easily the out-of-plane inclination of the tower along the different in-plane directions considered. Such inclination, as it will be explained later on in the paper, varies typically between about  $10^\circ$  to about  $2^\circ$ , passing from a direction almost parallel to y to the horizontal direction, respectively. Authors experienced that the SLP approach converges very quickly, with stabilization of the collapse inclination angle of the tilting plane after few iterations (3-5).

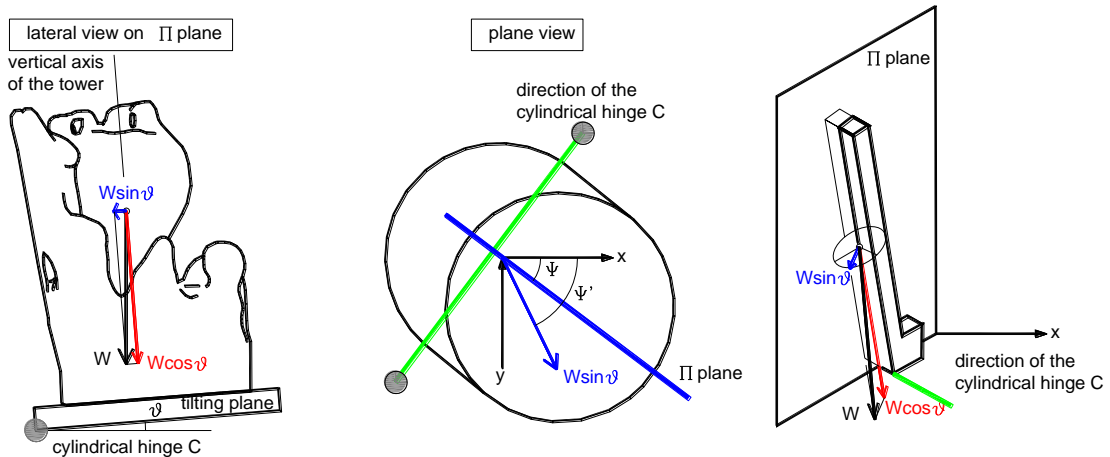


Fig. 7 –Limit analysis strategy for evaluating the collapse inclination of the 3D structure.

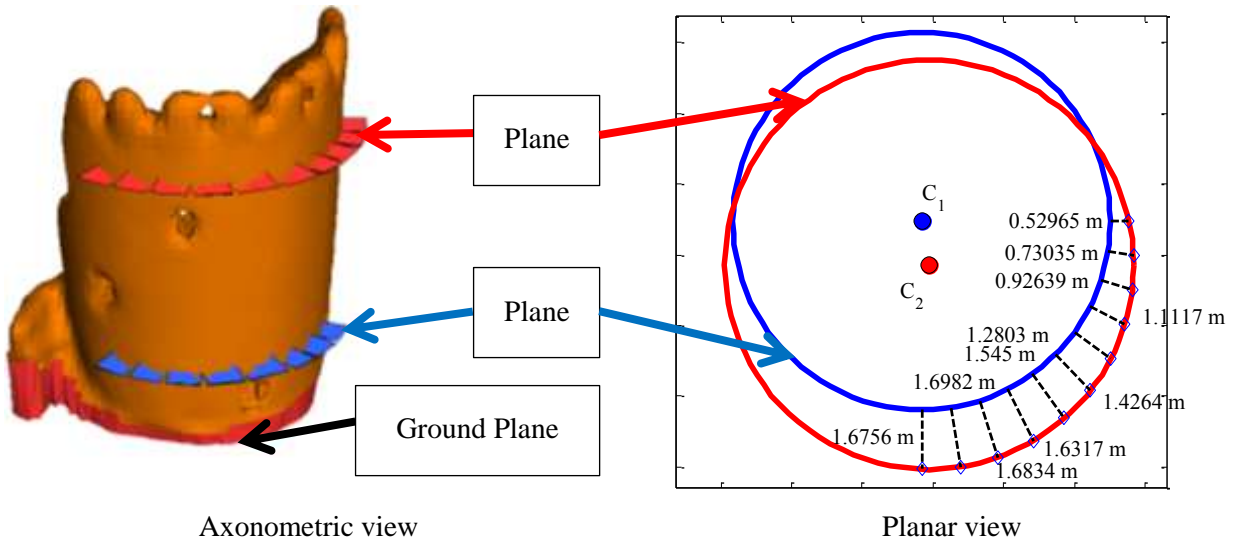


Fig. 8 – Evaluation of the actual inclination of the structure along different in-plane directions.

A final issue to tackle is the choice of the in-plan direction of the horizontal load, namely the angle  $\Psi'$  shown in Fig. 7. In general, the angle  $\Psi'$  cannot coincide with  $\Psi$ , the angle identifying the direction of plane  $\Pi$  (the angle  $90^\circ - \Psi$  identifies on the other hand the direction of the cylindrical hinge around which the tower rotates during the collapse under self-weight).

Authors experienced that it is necessary to scan angles  $\Psi'$  within the range  $-10^\circ$ - $100^\circ$  to obtain in output angles  $\Psi$  between  $0^\circ$  and  $90^\circ$ . Then, the identification of the  $\Psi$  associated with the minimum of the collapse inclination of the tower can be obtained by a simple spline interpolation of the different angles  $\Psi$  obtained by limit analysis computations.

#### 4 Results and discussion

Limit analyses are performed on the actual 3D geometry of the structure (Fig. 3h) which is derived from the laser scanning survey. The material of the tower, or more precisely all interfaces between adjoining tetrahedrons, ~~is~~was assumed to obey a Mohr-Coulomb criterion with tension cut-off, with mechanical properties approximating reasonably a continuum almost unable to withstand tensile stresses. In particular, considering that the tower is severally deteriorated and in agreement with consolidated literature [60], cohesion  $c$  and tensile strength  $f_t$  were kept constant through the tower and equal to 0.02 MPa. The assumption of small but non-zero values of cohesion and tensile strength improves the numerical stability of the linear programming solver and at the same time does not drastically increase failure multipliers. In such problems, indeed, almost all the stabilizing contribution is provided by gravity loads and internal dissipation turns out to play a negligible role on the increase of the collapse multiplier. Stability problems, such as the leaning historic masonry structures, are governed by geometry. Mechanical properties of the masonry tower were not available. Material properties assumed based on authors experience and from codes of practice. Thus, the friction angle  $\phi$  of masonry has been set equal to  $25^\circ$ , a value very similar to that assumed in the Italian code ~~in~~for the evaluation of the ultimate base-sliding shear for piers. Such value of the friction angle allows avoiding the formation of failure mechanisms due to sliding of macro-blocks, which are unlike in such kind of limit analysis problems. The choice to remove the cap in compression (see Fig. 5) has been made for a twofold reason: first of all, to limit as close as possible the optimization variables and second because the collapse of a tower after having reached the limit inclination angle occurs without evidence of crushing near the compressed toe. According to the two-step master-slave procedure proposed in the previous Sections, in Step 1 a rough linearization with 5 planes (and hence 5 plastic multipliers is adopted), whereas in Step 2 a piecewise linear

approximation with 25 planes (24 for the pure Mohr-Coulomb strength domain, 6 per quadrant and 1 for the tension cut-off) is assumed. The SLP procedure is here utilized to deal with the nonlinearity presented in such special limit analysis problem.

The resultant limit inclination angles of the structure so obtained varying the direction of the tilting plane are depicted in [Fig. 9](#) and compared with its actual inclination (thick black line), computed as shown in [Fig. 8](#). The collapse mechanism of the structure for a horizontal direction corresponding to the smallest limit inclination ( $\Psi = 60^\circ$ ) is highlighted in [Fig. 10](#).

From a careful analysis of the numerical results obtained, the following considerations can be drawn:

- 1) The tower is at present in a state not far from its collapse state. In particular, for a direction of the rocking plane  $\Pi$  roughly within the range:  $60^\circ \leq \Psi \leq 75^\circ$ , an additional inclination of  $1.5^\circ$  would lead to the collapse of the structure, see [Fig. 9](#). Considering the recent wetting-drying cycles which the structure has been subjected to, a careful analysis of the most suitable interventions to preclude rocking failure would be needed.
- 2) Crack pattern found by means of the limit analysis simulation corresponding to  $\Psi = 60^\circ$  ([Fig. 10](#)) shows the clear formation of a cylindrical rotational hinge near the base. The plasticization band is relatively narrow and tends to follow the geometric irregularities of the structure near the base, passing through the weakest transversal sections. The definition of a cylindrical hinge is therefore in principle not proper, because of both the finite thickness of the band and the not straight configuration of the plasticization zone.
- 3) The concentration of all the plastic flow near the base, in agreement with intuition, indirectly confirms that the master-slave approach proposed is fully consistent with the real behaviour.
- 4) Once the exact shape of the crack at the base is known, the failure mechanism is clearly identified and hand calculations (or assisted by a CAD program) can be performed to evaluate the position of the center of gravity of the macro-block subjected to rocking. In this way, the estimation of the collapse inclination angle is very straightforward and provides a further validation of the procedure proposed, as well as a ready to use instrument by common practitioners involved in the safety assessment of leaning historic masonry structures.

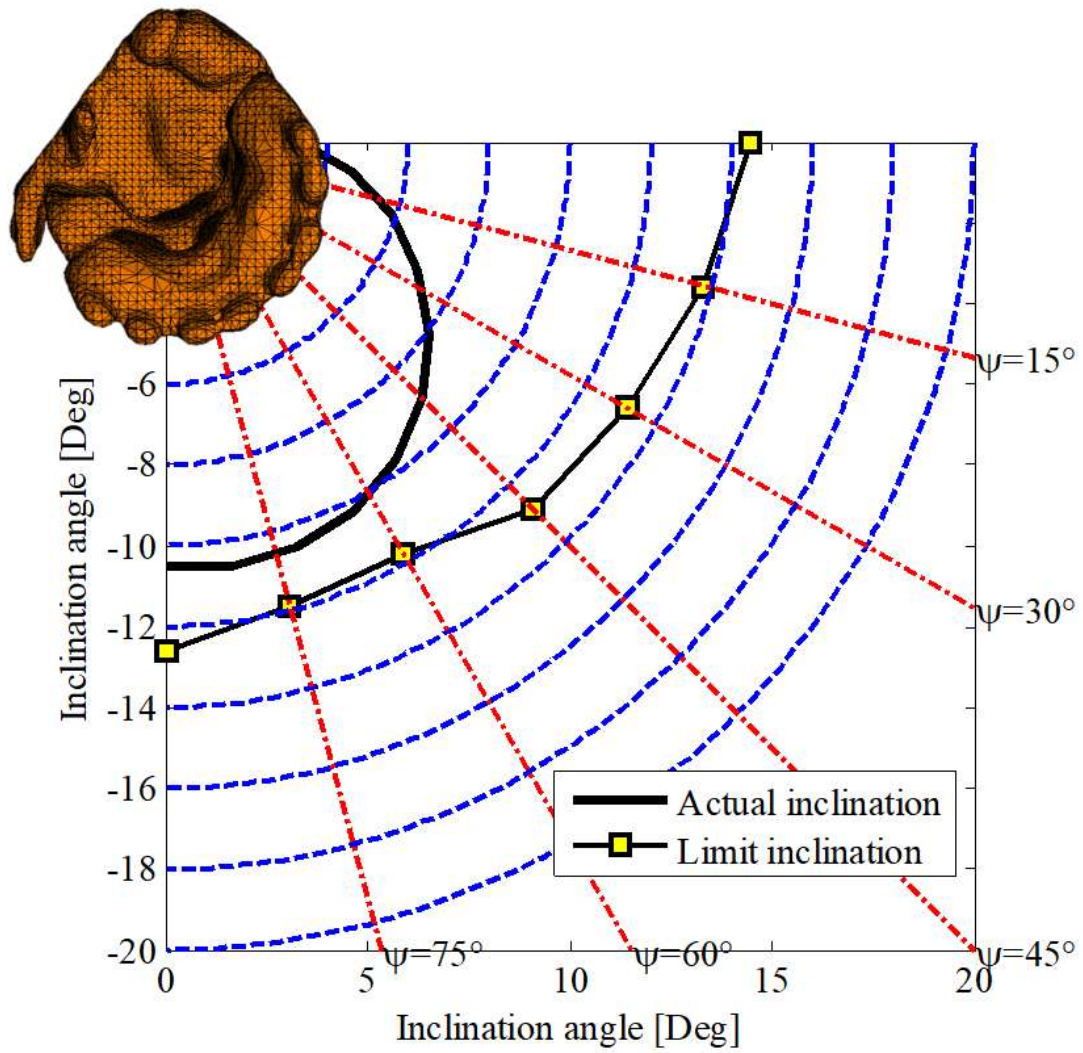


Fig. 9 – Comparison between actual inclination and limit inclination of the structure for several horizontal directions of the rocking direction.

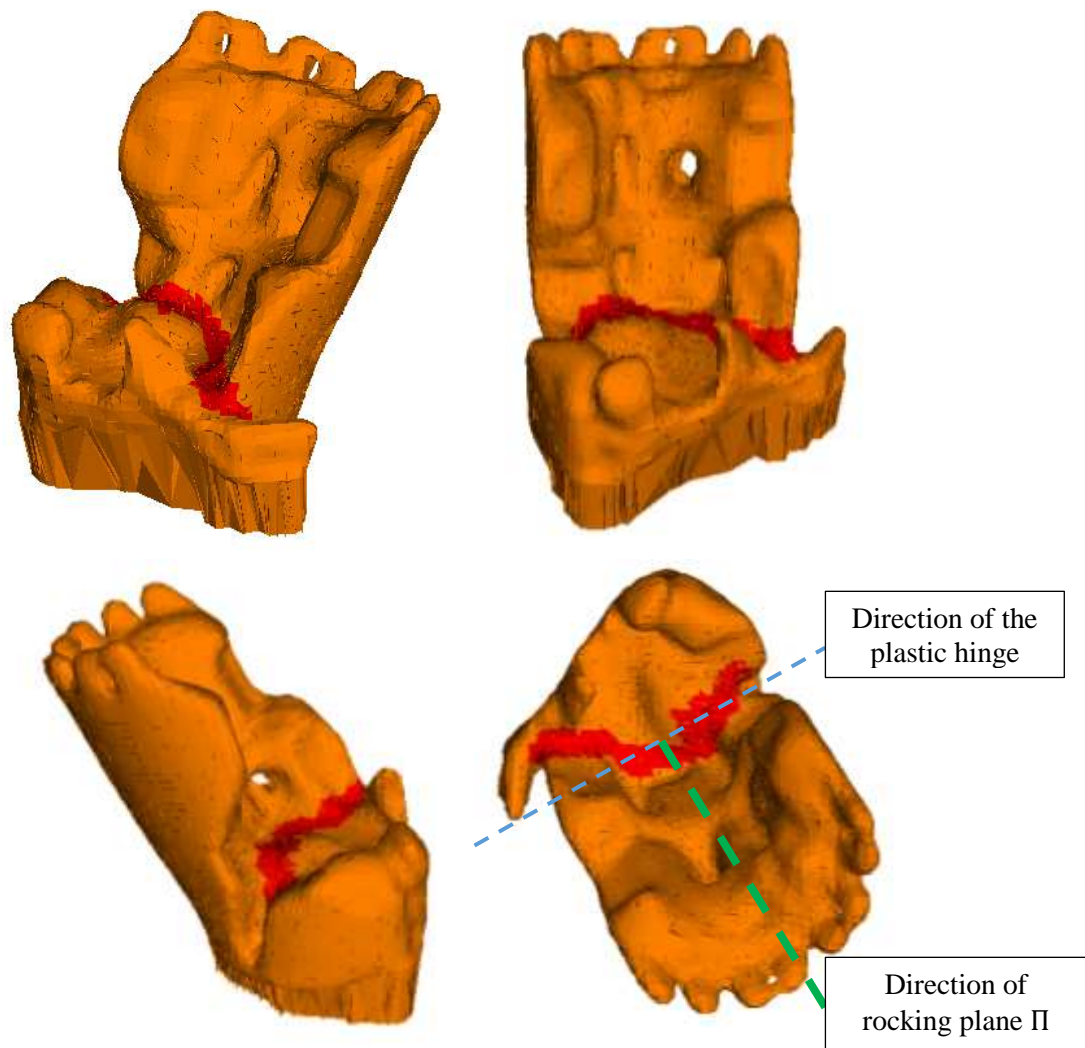


Fig. 10 – Collapse mechanism (with indication of the plastic processing zones) in the horizontal direction corresponding to the smallest limit inclination ( $\Psi=60^\circ$ ).

#### 4.1 Verification of the procedure: Enlarged active volume and single-step analysis

Additional limit analyses have been conducted to show the reliability of the assumption of rigid bodies (master-slave approach) by employing different sizes of the processing zones (Step 1), in order to check the convergence of the limit analysis solution (see Mesh #1 and Mesh #2 in [Fig. 11Fig. 11](#)). Particularly, Mesh #1 is characterized by an enlargement of the processing zone, with respect to the one employed in the previous sections (Present Mesh in [Fig. 11Fig. 11](#)), which represents the largest active volume faceable for the computers at disposal. Conversely, Mesh #2 ([Fig. 11Fig. 11](#)) represents a restriction of the previous processing zone, which also excludes interfaces where plastic dissipation occurred in the Present Mesh ([Fig. 10Fig. 10](#)). In addition, the mesh obtained through retopology (Coarse Mesh in [Fig. 11Fig. 11](#)) has been utilized to solve the LP problem in the framework of a single-step analysis, i.e. without master-slave approach.

[Fig. 12Fig. 12](#) shows the comparison among collapse inclinations of the structure at several horizontal directions of the rocking direction, with different meshes/processing zones tested. Furthermore, [Fig. 13Fig. 13](#) shows the comparison between the collapse deformed shapes of the coarse mesh (top) and Mesh #1 (bottom) in the horizontal direction  $\Psi=60^\circ$ . For the sake of brevity, the failure mechanism of Mesh #2, which is completely different from the others (characterized by a sub-horizontal crack surface), is not reported.

As can be noted in [Fig. 12Fig. 12](#), a very good agreement is achieved in terms of collapse inclinations between Mesh #1 and the Present Mesh. Conversely, Mesh #2 radically overestimates the collapse inclinations of the structure. Indeed, the excessive restriction of the active volume, i.e. the exclusion of interfaces where plastic dissipation occurs, is responsible for the increase of the load carrying capacity of the structure, as well as a different failure mechanism. Moreover, the collapse inclinations evaluated through the Coarse Mesh (single-step analysis), although slightly overestimated due to the coarser mesh adopted, are in a reasonable agreement with the ones obtained with Mesh #1 and the Present Mesh, confirming the reliability of the two-step approach proposed.

Additionally, the failure mechanisms collected in [Fig. 13Fig. 13](#) further validate the one obtained with the Present Mesh ([Fig. 10Fig. 10](#)). Indeed, they substantially show the same failure mechanism with plastic dissipation essentially in the same zones (compare [Fig. 13Fig. 13](#) with [Fig. 10Fig. 10](#)), although the failure mechanism obtained with the Coarse Mesh ([Fig. 13Fig. 13](#), top) is computed through a single-step approach (fully active volume). Therefore, this outcome further confirms the effectiveness of the master-slave approach proposed and the choice of the processing zone adopted for the Present Mesh ([Fig. 6Fig. 6](#)).

Finally, [Table 1Table 1](#) collects the computational times required to solve the limit analysis problems. The following considerations can be drawn:

- 1) The time needed to solve the problem with the Present Mesh is around 18 hours, considerably less than the time needed by Mesh #1. Therefore, the Present Mesh, which is more accurate than the Coarse mesh ([Fig. 11Fig. 11](#)), appears a suitable compromise to speed up computations with a precise geometric and mechanical approximation of the real situation.
- 2) Step 2 requires generally more time to be performed, because of the utilization of several planes to approximate interfaces failure surface. Mesh #2 requires the same time for Step 1 and Step 2 because of the too restricted processing zone in Step 1, which does not allow to find a plausible failure mechanism.



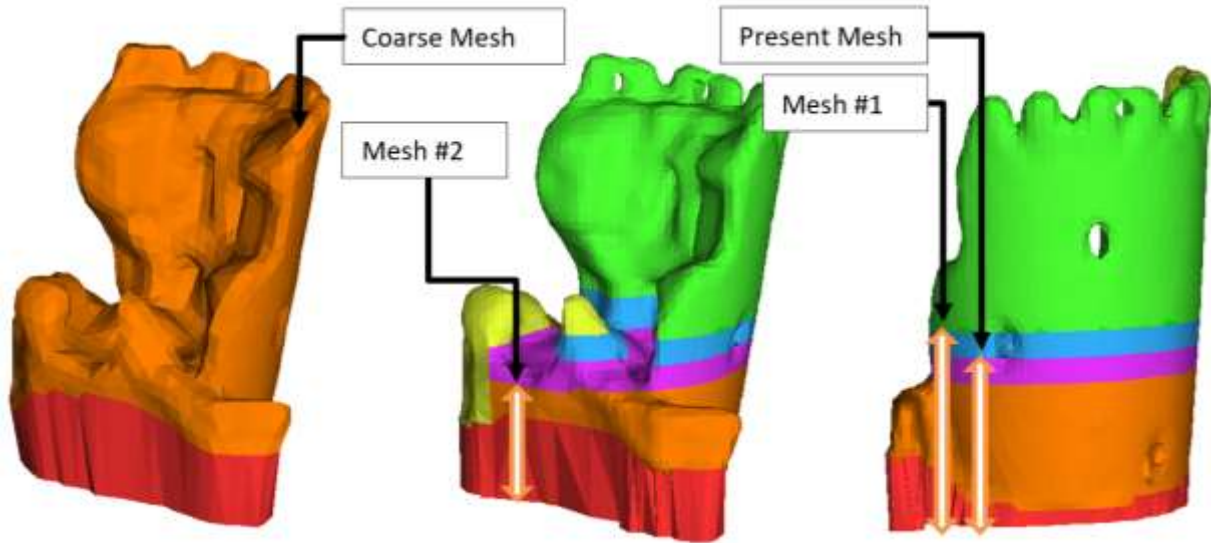


Fig. 11 – Different sizes of the processing zones (Step 1) tested to check the convergence of the limit analysis solution (middle and right) and coarse mesh (left) utilized to solve the LP problem (single-step approach). Present Mesh refers to the processing zone depicted in [Fig. 6](#).

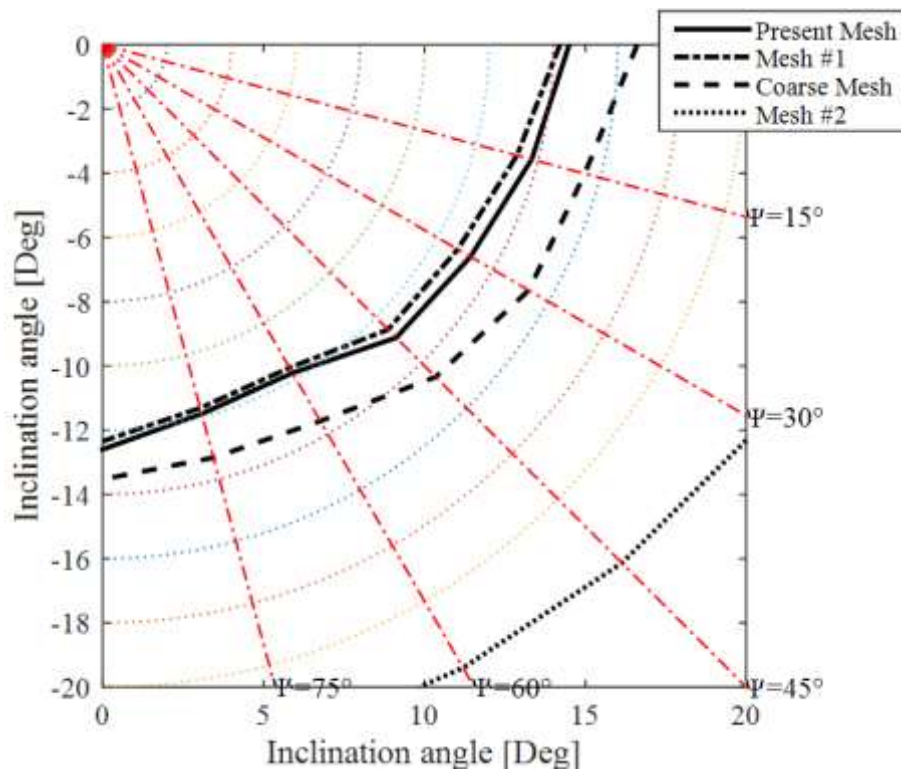


Fig. 12 – Comparison among collapse inclinations of the structure at several horizontal directions of the rocking direction and with the different meshes/processing zones tested.

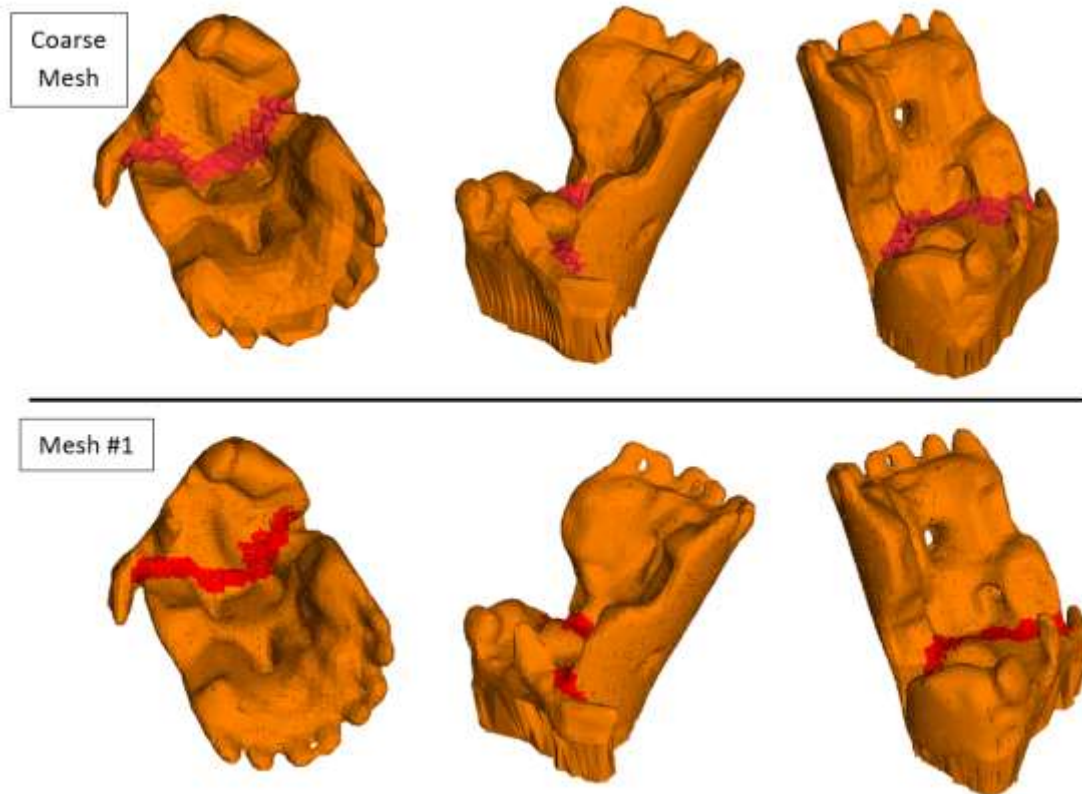


Fig. 13 – Comparison between collapse deformed shapes (with indication of the plastic processing zones) for the coarse mesh (top) and Mesh #1 (bottom), in the horizontal direction corresponding to the smallest limit inclination ( $\Psi=60^\circ$ ).

Table 1. Times required to solve the limit analysis problems.

	Solution of the LA problem <sup>(x)</sup> hh:mm:ss	
	Step 1	Step 2
Present Mesh	06:12:44	12:01:05
Mesh 1	15:48:09	33:19:52
Mesh 2	03:31:28	03:27:02
Coarse Mesh (single-step analysis)	21:25:13	

<sup>(x)</sup> CPLEX solver on a work station equipped with 64bit 2x16GB DDR4 RAM (16 slots). Computations are referred to the evaluation of the collapse inclination for 7 tilting directions. Computations include also pre-processing phases, such as identification of the interfaces and assemblage of the equality and inequality constraints.

#### 4.2 Verification of the procedure: Nonlinear finite element analysis

A further verification of the proposed structural analysis approach is achieved by means of the comparison with the results of a nonlinear finite element tilting plane incremental analysis. In this analysis, isotropic plastic-damage behaviour [61] is adopted for masonry. Although masonry is a complex material typically characterized by anisotropic behaviour, the hypothesis of isotropic nonlinear material generally appears suitable for historic masonries, as they usually present chaotic and random textures [33, 32, 28].

The adopted continuum plastic-damage model [61, 49] is characterized by a yielding function with multiple-hardening variables and two independent scalar damage variables, one for the tensile damage  $d_t$  and the other for compressive damage  $d_c$ . Being the model formulated in the context of nonassociated plasticity [61], the plastic potential is defined by the dilatancy angle  $\psi$ , generally assumed equal to  $10^\circ$  for masonry [62], as well as by a smoothing parameter  $\epsilon$  usually assumed equal to 0.1 [63]. In addition, the strength domain is specified by the ratio  $f_{b0}/f_{c0}$  between the biaxial  $f_{b0}$  and uniaxial  $f_{c0}$  initial compressive strengths, typically assumed equal to 1.16 [64], and by the shape constant  $\rho$ , normally assumed equal to  $2/3$  [63].

Reference to the Italian code has been made to set the mechanical properties of the material (cluttered stone masonry), which are collected in [Table 2Table 2](#). The tensile strength has been kept equal to the value used in the FELA, whereas the evolution of the scalar damage variables  $d_t$  and  $d_c$  has been kept substantially proportional to the decay of the uniaxial stresses, as adopted in several numerical campaigns [32, 63, 33] ([Table 2Table 2](#)).

Dead load is initially applied to the structure through an incremental procedure, considering clamped boundary conditions at the base. Then, a pattern of imposed displacements, which simulates a tilting plane, is incrementally applied at the base of the structure. Abaqus Standard [49] has been used to conduct the simulation. Geometric nonlinearity has been considered to account for large-displacement effects, which, in this case, are expected to play a fundamental role.

[Fig. 14Fig. 14](#) shows the tensile damage ( $d_t$ ) contour plot obtained with a nonlinear finite element tilting plane incremental analysis. In particular, [Fig. 14Fig. 14](#) refers to the condition in which the base inclination of the tower equals the limit inclination computed through FELA, in the horizontal direction with the smallest limit inclination ( $\Psi = 60^\circ$ ). As can be noted, the crack pattern in [Fig. 14Fig. 14](#) is in good agreement with the collapse mechanism observed with FELA ([Fig. 10Fig. 10](#) and [Fig. 13Fig. 13](#)). Indeed, all of the failure mechanisms are governed by the overturning of the highest part of the tower with a detachment from the remaining part, which pseudo-horizontally runs from one side of the tower to the opposite one. Although the crack pattern in [Fig. 14Fig. 14](#) also shows a pseudo-vertical crack along the central part of the tower's trunk, it does not cross the thickness of the wall ([Fig. 14Fig. 14](#)). Therefore, such crack does not appear essential in the main failure mechanism of the tower. Rather, this vertical crack develops after the formation of the horizontal main fissure, being the result of large-displacement effects.

Finally, it should be pointed out that the evaluation of the limit inclination angle by means of the standard nonlinear finite element tilting plane analysis herein adopted is non-trivial and, in facts, requires specific interpretation of the results. Therefore, further studies on this topic are needed. Consequently, the proposed FELA approach appears more robust than standard nonlinear FE methods, being able to directly compute the limit inclination angle of a masonry structure.

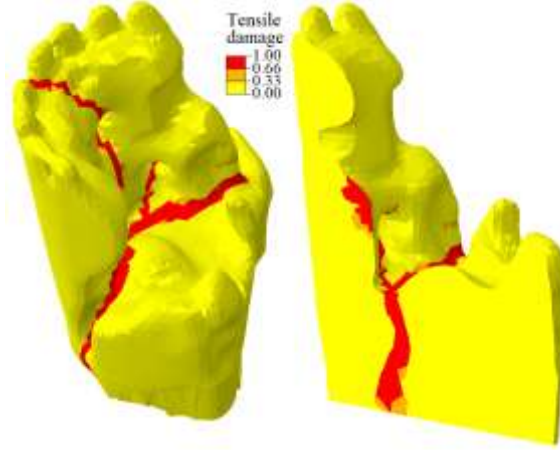


Fig. 14 – Tensile damage contour plot obtained with a nonlinear finite element tilting plane analysis, in the condition of a base inclination of the tower equal to the limit inclination computed through FELA, in the horizontal direction with the smallest limit inclination ( $\Psi = 60^\circ$ ).

Table 2. Mechanical properties adopted for masonry in the FELA and in the continuum plastic-damage models.

FELA model						
Tensile strength $f_t$ [MPa]		Cohesion $c$ [MPa]		Friction angle $\phi$		
0.02		0.02		25°		
Continuum plastic-damage model						
Young's modulus[MPa]	Poisson's ratio	Density [kg/m3]	$\epsilon$ [°]	$\psi$ [°]	$f_{b0}/f_{c0}$ [°]	$\rho$ [°]
870	0.15	1900	0.1	10°	1.16	2/3
Tensile uniaxial nonlinear behaviour			Compressive uniaxial nonlinear behaviour			
Stress [MPa]	Inelastic strain	$d_t$ [°]	Stress [MPa]	Inelastic strain	$d_c$ [°]	
0.02	0	0	1.0	0	0	
0.001	0.0001	0.95	1.1	0.001	0	
			0.05	0.007	0.95	

## 5 Conclusions

In this paper, a simplified and rapid procedure for the automatic transformation of point clouds (surveyed on historic structures) to 3D FE meshes, passing from the concept of watertight mesh, and their utilization in a multi-step upper bound limit analysis with automatic variables reduction, has been proposed. The level of approximation introduced in the actual geometry appeared to be included within the engineering tolerance.

Following the Heyman's intuition, the use of 3D FELA on the generated mesh is implemented to evaluate the critical condition (i.e. maximum inclination capacity) of a leaning historic masonry structure. In this way, the structural health condition of a historic structure is evaluated by comparing the maximum critical inclination angle against the current one. Considering that FE meshes obtained from detailed laser scanner surveys would consist of many variables to make the limit analysis impossible to solve even with powerful workstations, a recursive identification of the processing zone has been proposed, excluding from computations all those elements which do not undergo plastic deformation. The automatic procedure adjusts iteratively the processing zone, progressively restricting the analysis to the few elements interested by the failure mechanism and

considering the other portions outside the processing zone as rigid blocks. Also, a sequential linear programming kernel has been adopted to linearize the normalization condition, which becomes nonlinear if the inclination angle at failure is considered as collapse multiplier.

To demonstrate the effectiveness of the automated procedure, the southwest leaning ruined tower of the Caerphilly castle (Wales, UK) has been employed as a case study. It emerged that the tower in its actual condition is not far from its collapse. Indeed, an additional inclination of the structure by  $1.5^\circ$  appears to be critical. In this limit condition, the collapse mechanism of the structure has been found in agreement with intuition, i.e. it consists in the overturning of the main leaning part of the structure. The results have been further validated by means of the comparison with the outcomes of single-step analyses (fully active volume) and nonlinear FE analyses.

The procedure proposed herein is characterized by a high degree of automation at each operational level. Such approach could be effectively utilised to assess the stability of historic structures at a national scale and provide useful information to engineers and managers to classify the structural health condition of historic assets in their care.

Although the procedure proposed represents a novel solution for evaluating the stability of extant masonry structures, it could be enriched with an adaptive mesh enhancement in the framework of a multi-step strategy.

### Acknowledgements

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